## Using computers to go where fluid dynamics experiments cannot

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## Outline

- Heat transfer analysis in internal turbine cooling

- Passive scalar separation using chaotic advection



## AM Motivation - From Flying and Thermo



Figure: http://www.milnet.com/jeteng.htm

## A Motivation - From Flying and Thermo



Figure: Brayton cycle http://grc.nasa.gov
Efficiency as a function of temperature ratio: $\eta_{\text {cycle }}=1-\frac{T_{5}}{T_{4}}$ Increase $T_{4}$, Limits: Metal melting temperature and part life

## AIM <br> Transient hot spots can cause part failure



Figure: Turbine blade leading edge region, Right: from Langston (1980)

## $\sin$ <br> Stagnation region horseshoe vortex is unsteady



Figure: PDF measurements from Radomsky et al. (2000)

## A highly resolved LES simulation is proposed

The large scales are solved on the grid while subgrid scales are modelled.

$$
\begin{align*}
\nabla \cdot \mathbf{U} & =0,  \tag{1}\\
\partial_{t} \mathbf{U}+\mathbf{U} \cdot \nabla \mathbf{U} & =-\rho^{-1} \nabla P+\nabla \cdot\left(\left[\nu+\nu_{t}\right] \nabla \mathbf{U}\right),  \tag{2}\\
\partial_{t} T+\mathbf{U} \cdot \nabla T & =\nabla \cdot\left(\left[\alpha+\alpha_{t}\right] \nabla T\right), \tag{3}
\end{align*}
$$

- Initial estimates based on a steady RANS computation (Knost et al. 2009) at $R e_{C h o r d} \approx 150,000$ :
- $10^{8}$ cells (for $x^{+} \approx y^{+} \approx z^{+} \approx 50$ )
- $10^{6}$ time steps per flow through (based on CFL)
- Highly scalable, open-source Spectral Element code

$E=3, N=4$
Fischer et al. 2007

- Strong scaling for 7.8 mio grid points


## $\widehat{A} \mid \vec{M}$ Inflow boundary



- 2D-periodic, divergence free solution of Navier-Stokes (Taylor vortices)


## Ell <br> Side domain view








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－Increased grid density near wall
－Length scale by grid spacing
－Freestream intensity by inflow vortex strength
－Boundary layer by slope and length of converging section

## A Preliminary results


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## $\overrightarrow{\mathbf{A}}$ Preliminary results


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## A Preliminary results



## $\overline{A N}$ <br> Efficient Mixing in Laminar Flows Through Chaotic Particle Trajectories

- Exponential stretching of interface across which diffusion occurs
- Can be generated from simple flow fields.



## $E \pi$ <br> Braiding with "Ghost-Rods"

Stirring in a braiding motion with physical rods

P. L. Boyland, H. Aref, and M. A. Stremler, "Topological fluid mechanics of stirring," J. Fluid Mech., 2000

Physical rods replaced by periodic orbits

$$
u=\frac{\partial w}{\partial y}= \pm \sum_{n=1}^{N} U_{n} \sin (n x / 2)
$$



$$
u=\stackrel{\partial \psi}{\partial y}=\mp \sum_{n=1}^{N} U_{n} \sin (n x / 2)
$$


"Stirring with ghost rods in a lid-driven cavity," by Pankaj Kumar, Jie Chen, and Mark Stremler.

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## $A$

The Chebychev-Fourier Method was used to solve $\left[F T^{2}\right.$ ?
a Vorticity-Stream Function formulation a Vorticity-Stream Function formulation
source: Roger Peyret, Spectral Methods for Incompressible Viscous Flow, 2002



Contours of Stream Function

## A Contour Plots of Stream Function




(b) $R e=1$

(c) $\mathrm{Re}=10$

(d) $\mathrm{Re}=100$

## A Mixing Index for Passive Scalar Transport



## A Mixing Index for Passive Scalar Transport



ReSc $=10,000$
Mixing Index:

$$
M=\frac{1}{N} \sum_{i=1}^{N} \frac{\theta_{0}-\left|\theta_{i}-\theta_{0}\right|}{\theta_{0}}
$$


$\operatorname{Re}=1, t=1.00[s]$

$\begin{array}{lllll}0.1 & 0.2 & 0.3 & 0.4 & 0.5\end{array}$


## A Dispersion of Particles



## A $\mathbb{M}$ Stirring Index for Different Re Based on the Box [ $\left.F]^{1}\right]^{2}$ Counting Method


(a) Stirring Index 1

(b) Stirring Index 2

Stirring Index: $\epsilon=\frac{1}{K} \sum_{i=1}^{K} \omega_{i}$

$$
\omega_{i}= \begin{cases}\frac{n_{i}}{n_{\max }} & , \quad n_{i}<n_{\max } \\ 1 & , \quad n_{i} \geq n_{\max }\end{cases}
$$

## Comparison of Dispersion of Particles between $\operatorname{Re}=0.1$ and $\operatorname{Re}=10$


(a) $\mathrm{Re}=0.1,8$ Advection Cycles
(c) $\mathrm{Re}=0.1,18$ Advection Cycles

(a) Contours of $\theta_{1}$

(C) Contours of $\theta_{1}$ after unbraiding

(b) Contours of $\theta_{2}$

(d) Contours of $\theta_{1}-\theta_{2}$

## $\operatorname{Aln}$ <br> Chaotic Separation vs. diffusion



## $A \mathbb{A}$ Using computers to go where fluid dynamics $\left[F \lambda^{2}\right.$



Texas A\&M Supercomputing Center has played an important role in this work.

